

*2022-2023 Performance Report of Tohoku University Students' Projects
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Project for real-time control of flow field using sparse processing PIV and plasma actuators

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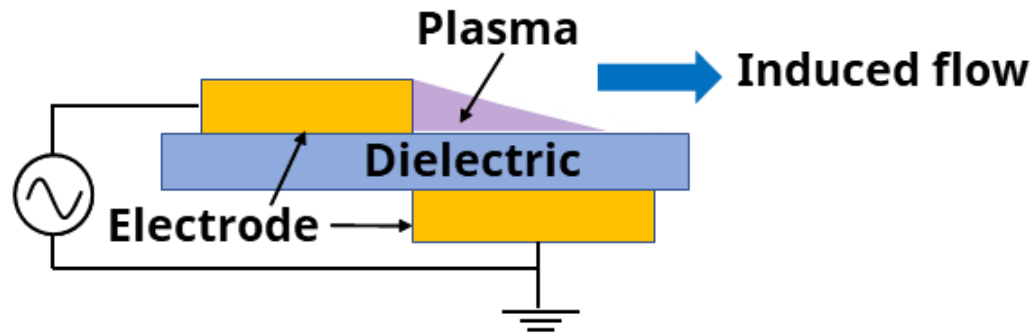
□ Various fluid machines

- Airplanes, drones, automobiles, windmills...
 - Expectations for performance improvement in active flow control



□ Plasma actuator (PA) (Corke et al., 2007, Fujii, 2014)

- Induced flow by plasma

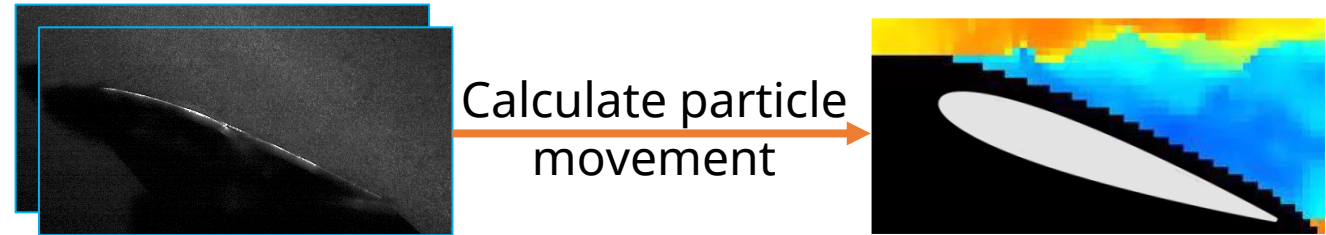


- Optimal control input depends on airflow conditions

➔ **Feedback control** according to flow field is important

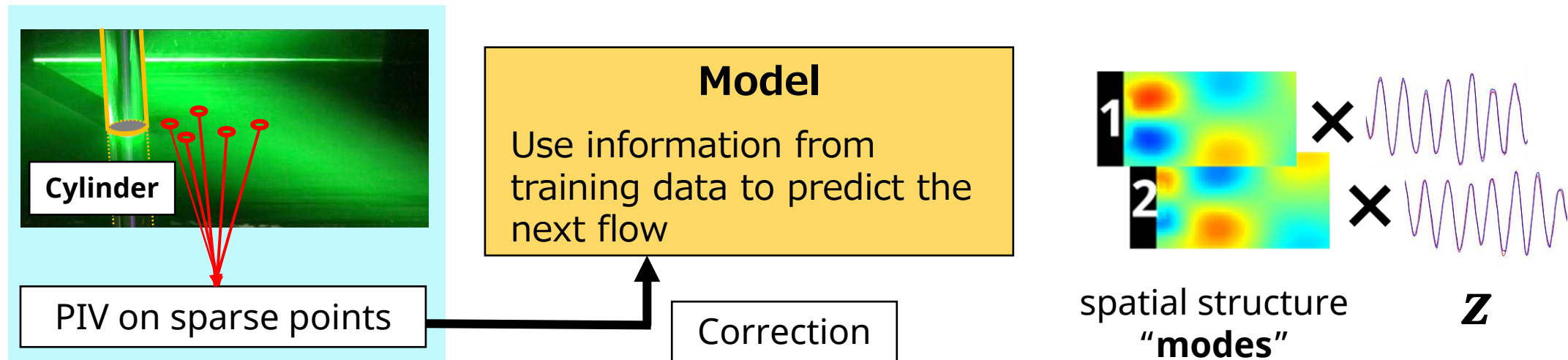
□ How to measure the current flow with high accuracy?

- Particle image velocimetry (**PIV**)
- The flow field data is huge



□ Sparse Processing PIV (**SPPIV**) (Kanda *et al.*, 2021, 2022)

- Estimating the overall state with information from a sparse points (= sensors)
- Estimate Z (the "intensity" of the spatial structure known in advance)



Construct a real-time feedback control system using SPPIV and PA and evaluate the estimation performance and control effect

Target

Kármán vortex suppression

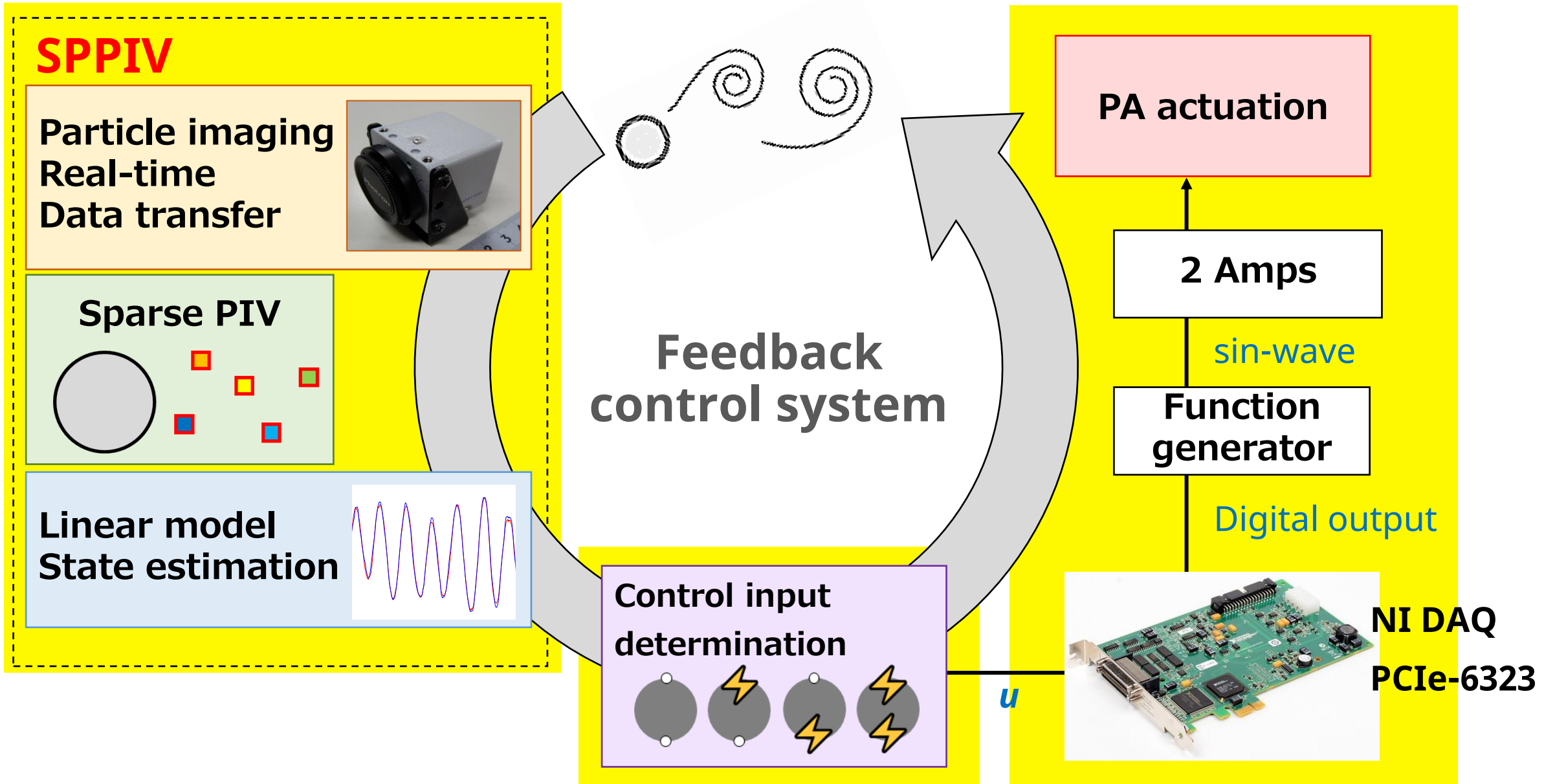


□ Construct binary control rules by turning PA ON/OFF

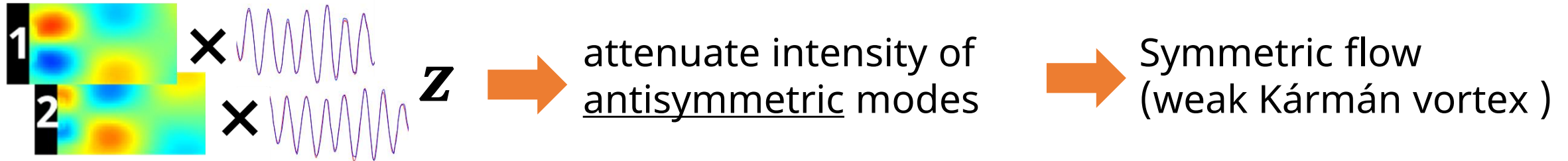
- One-point predictive control
- Advanced control (control in section)

□ Feedback control in wind tunnel testing

- Evaluation of control effect by focusing on symmetry of flow velocity field



□ Construction of feedback control rules

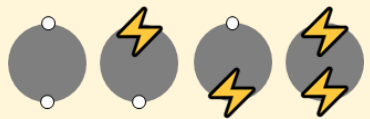


➤ Control law 1

One-point Predictive Control

- Predict z_{k+L} of each u_k
- Consider the next one step

Calculation for
all patterns



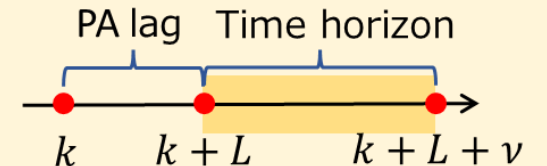
$$\operatorname{argmin}_u \|z\|_2$$

➤ Control law 2

Advanced control (Ikeda *et al.*, 2017)

- Consider the interval between ν steps

$$(v = [u_k^T \ u_{k+1}^T \ \dots \ u_{k+\nu-1}^T]^T)$$



- Solve the optimization problem

$$\operatorname{argmin}_v (\|z_{k+L+\nu}\|_2 + \text{binarize})$$

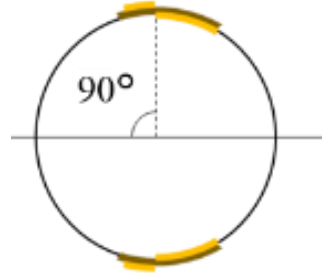
7 Experiment Setup

□ Test equipment

- Wind tunnel : Tohoku-University Basic Aerodynamic Research Wind Tunnel (T-BART)
- Camera : IDP-Express R2000 (Photron)
- Airfoil section : Cylinder model ($d = 30\text{mm}$)

□ Test condition

- Flow speed : 5 m/s
- Sampling rate : 4000 Hz
- Re : 1.1×10^4

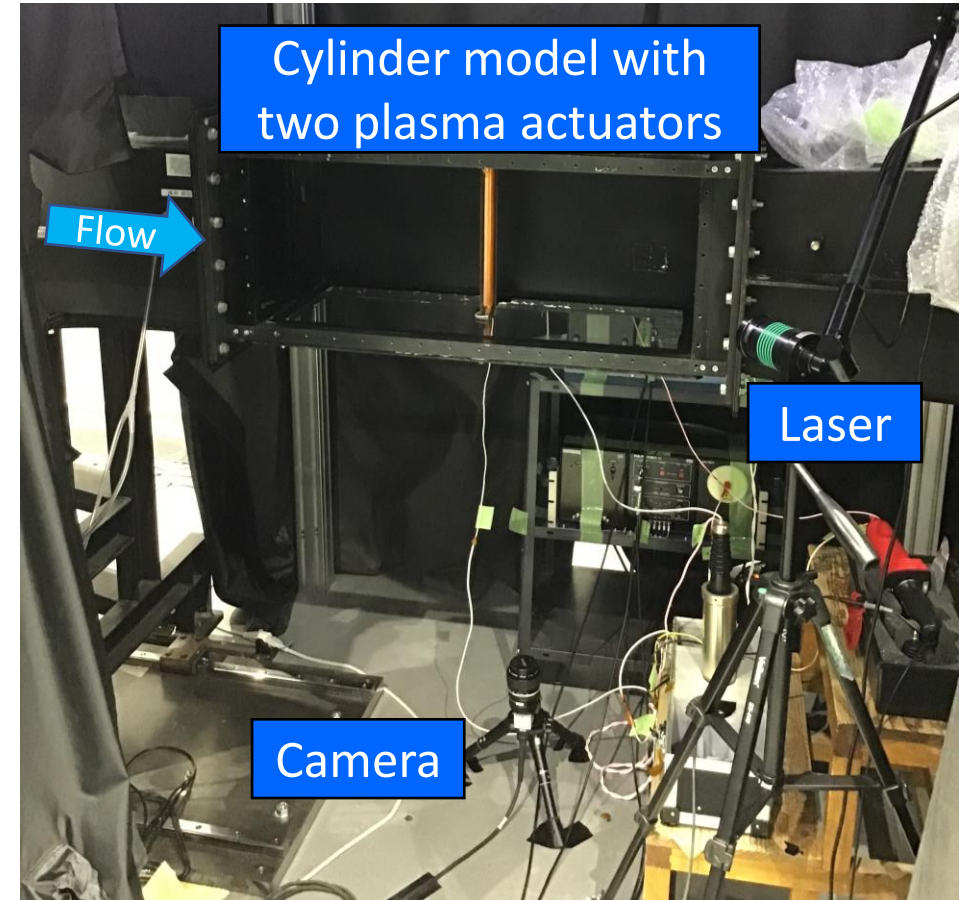


□ Plasma actuator

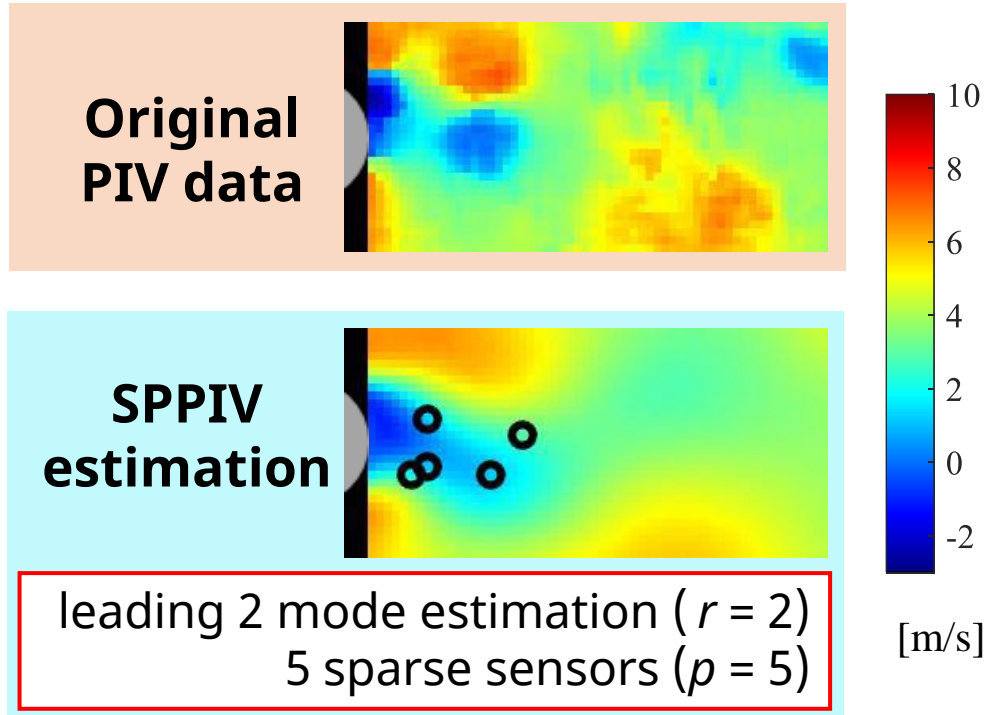
- waveforms : 8kVpp, 10000Hz sin-wave
- Switching ON/OFF
for training data : **100Hz random**

□ Analysis

- Estimation modes r : 2
- Number of sensors p : 2, 5, 10
- Parameters of l_1 opt. : $\nu = 40$ (Time horizon), $\alpha = 0.5$ (weight parameter)

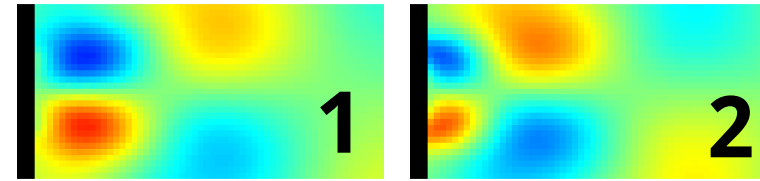


□ Velocity field (x-direction)



- Tracks large structures well
- Expects upper 2 modes to be sufficient for control

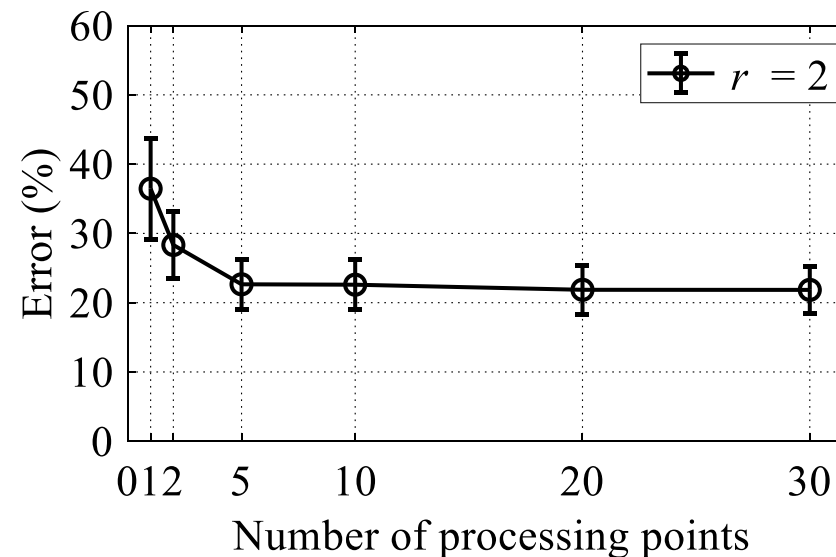
□ Spatial modes for reconstruction



Leading two modes represent **>50%** (Energy-based)

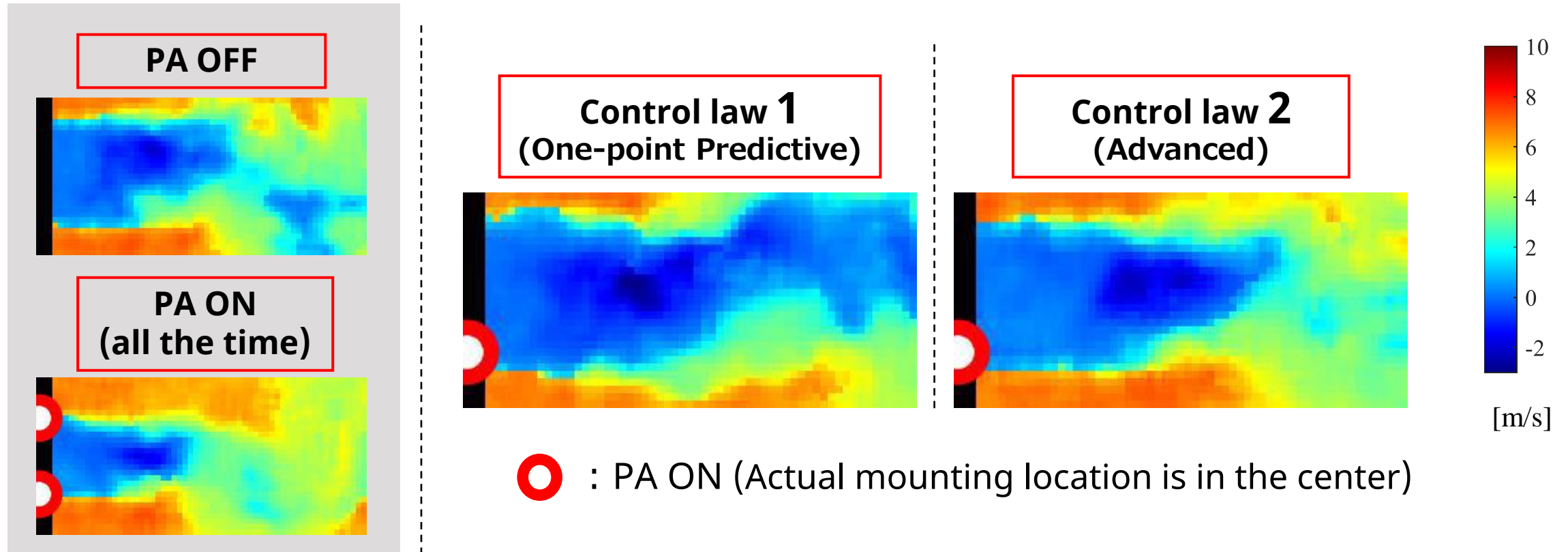
- Two modes represent the periodic movements

□ Estimation error



- High accuracy
- 5 sensors are enough

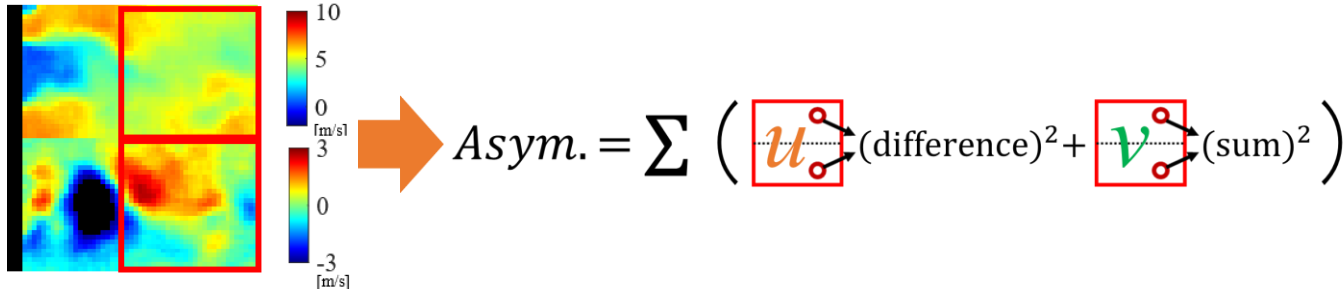
□ Velocity field in x-direction (PIV was repeated for all pixels of flow field)



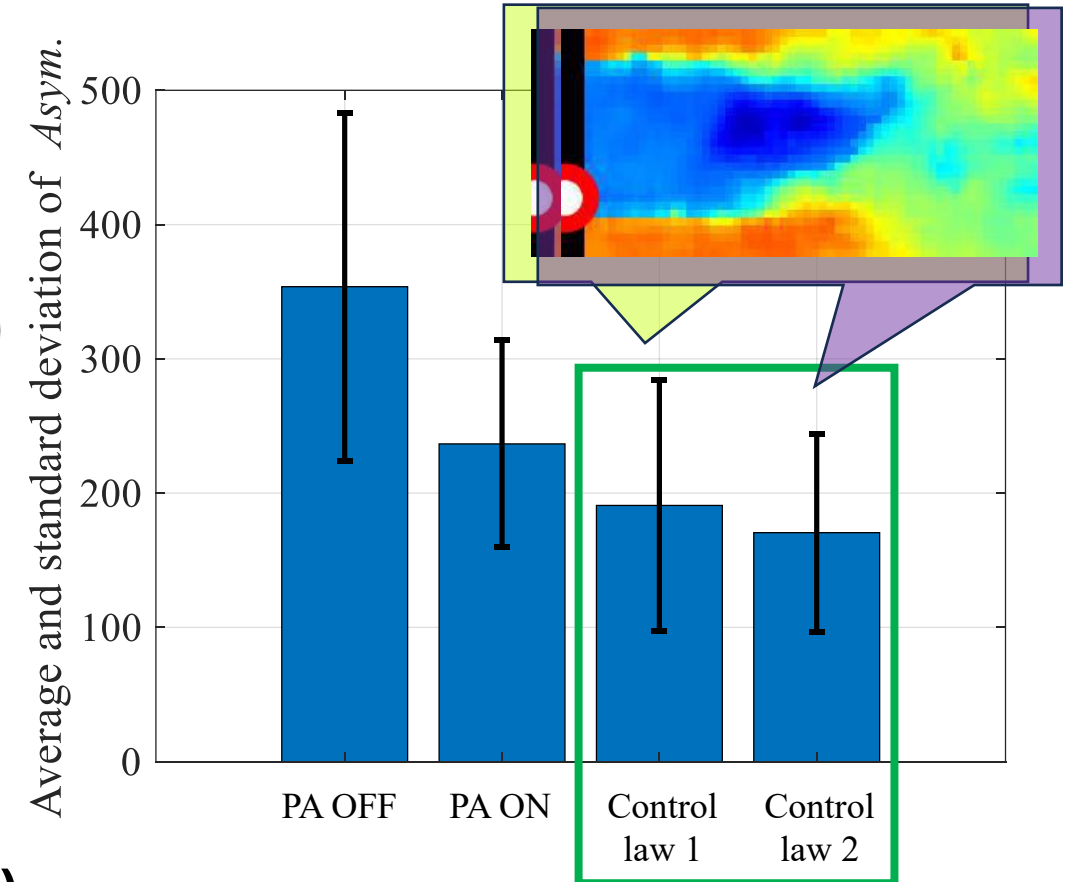
- The large-scale structure was disturbed by switching PA ON/OFF
- Time period when Karman vortex is suppressed exists (not continuous)
- **Control law 2** uses two PAs discharging at the same time

□ Evaluation of control effectiveness

- Evaluated by asymmetry of the flow field



- Certain control effect confirmed
- **Law 2** : More stable with small variance
→ Due to time horizon



□ Processing time by step (5 sensors, average)

	Law 1	Law 2
SPPIV + determine u	60 μs	97 μs

- Short enough for a sample rate of 500 μs
- **Law 2** : ADMM enables high-speed iterative calculations

*Alternating Direction Method of Multipliers

Constructed a real-time feedback control system using SPPIV and PA and evaluated the estimation performance and control effect

□ Results

- Real-time feedback control is achieved through wind tunnel testing
- It is found that control by time horizon considerations is better

□ Future works (for my master's degree)

- Parameter tuning for feedback control data
- Improving training data construction process